



EDA = 6.0 deg

8.5"

0.5 deg  
washout  
twist

7.5 deg

13"

1.0"

2.9 deg

18"

Re sqrt(CL) = 70K  
(85K)

2 layers

65K  
(80K)

1 layer

50K  
(65K)

35K  
(45K)

Design Loads

wing lift 150 lb @ 50 m/s  
 root bend.mom. 1400 lb-in  
 root cap load 3300 lb  
 root cap area 0.037 in^2 top  
 0.030 in^2 bot  
 cap stress 89 ksi top  
 110 ksi bot  
 wrap stress 12 ksi (shear)  
 root torsion 25 lb-in  
 root skin th. 0.015 in  
 wing skin shear 290 psi

Wing spar weights

caps 60 g (prepreg uni CF)  
 core 15 g (6 lb endgrain balsa)  
 wrap 14 g (2 oz glass)  
 join 15 g (0.315" CF tube)  
 tubes 3 g (Kevlar)  
 boxes 4 g (plywood, glass)  
 glue 4 g

Wing structure weights

core 80 g (2.3 lb Spyderfoam)  
 skin 90 g (1.7oz Kevlar)  
 spar 115 g  
 fill 6 g (microballoons+flox)  
 webs 6 g (vertical grain balsa)  
 misc 15 g

Wing panel weights

center 210 g  
 tips 122 g

CF boom  
 0.59" -> 0.34" ID  
 0.016" wall

2 layers

**Aegea 2m**

Mark Drela 24 May 02

mass = 21 oz  
 m/A = 6.2 oz/ft^2  
 area = 490 sq in  
 span = 78.6 in  
 A.R. = 12.6

Camber Settings

	Flap	Aileron
winch:	+20	+20
zoom :	-2	-2
run :	-2	-2
range:	0	0
float:	+3	+2
brake:	+70	+10

Kevlar fuselage shell  
 1.7oz 4 layers,  
 CF reinforcements

wing +2 deg.

HT12  
 5.0% t  
 0.0% c

HT08  
 5.0% t  
 0.0% c

AG47ct  
 5.0% t  
 1.7% c  
 1.3% c  
 -0.5 deg

Spyderfoam  
 balsa insert  
 2.9 oz CF caps  
 1.0 oz Kevlar

All-moving tail  
 49 sq in 10.0%  
 Vh = 0.44  
 -17 ... +20 deg

AG46ct  
 6.1% t  
 2.0% c  
 1.7% c  
 -0.5 deg

Item weights

650 NiMH 53 g  
 4 JR241 36 g  
 1 HS85MG 24 g  
 Hitec 555 23 g  
 wiring 10 g

fuse 78 g  
 boom 17 g  
 stab 10 g  
 rudd 8 g  
 wing 332 g

AG45ct  
 6.9% t  
 2.2% c  
 1.8% c

AG44ct  
 7.3% t  
 2.3% c  
 1.9% c

36 sq in 7.3%  
 Vv = 0.029  
 +/-30 deg

CF V-strut  
 CF platform/horn  
 4-40 Nylon bolt

Spyder foam  
 1.0 oz Kevlar  
 Kevlar hinge

tail 0 deg.

